
NASA's Mission Operations and Communications Services

**This Description applies only to proposals in response to
NASA's Announcement of Opportunity for Discovery Missions**

AO 00-OSS-xx

May 2000

This document is intended to assist in the preparation of proposals in response to the Announcement of Opportunity (AO), AO-00-OSS-X issued by NASA's Office of Space Science (OSS) for Discovery Missions. NASA provides many operations and communications services that are available for NASA missions. The use of these services will incur costs to the user and estimates for these costs need to be included in proposals submitted under this AO. To facilitate proposal preparation, proposers are encouraged to read this appendix and contact the individuals named in Section 1.5 below.

As a matter of policy, NASA will include estimated costs for mission operations and communications services, as well as an assessment of key parameters for mission operations, in the evaluation and selection processes for all Earth-orbiting and deep space missions. The OSS and the Space Operations Management Office (SOMO), are implementing this policy:

- ## 1.2 Choice of Service Providers

Proposers are free to use all, some, or none of the NASA-provided services described below. Regardless of their choice, the proposal must include a rationale for the level of communications and mission services proposed and the costs of these services. **Required services should be identified irrespective of the provider.** Key communications and mission services parameters are listed below.

As a matter of policy, proposers should be prepared during Phase B to support tradeoff studies with the OSS and SOMO on the use of NASA-provided services versus proposed alternatives. In general, NASA-provided services should be employed whenever they meet mission objectives at a cost that is less than or equal to any proposed alternatives.

The Deep Space Network (DSN) is available for use by Discovery missions. All Discovery missions selected to date have employed or intend to employ the DSN as primary means of communicating with the spacecraft in flight. However, it is important to note that Discovery missions are not required to use the DSN.

SOMO is responsible for the functional management of most NASA space operations facilities. SOMO is the NASA program manager for the Deep Space Network (DSN), other NASA Ground Network (GN) stations, the Tracking and Data Relay Satellite System (TDRSS), the NASA Information Services Network, and certain mission operations facilities located at NASA centers.

1.4 SOMO Service Categories

Table 1-1: SOMO Service Categories

1.5 Process for Requesting Services

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The primary SOMO point of contact for this AO is the SOMO Lead Center Customer Commitments Manager (CCCM) for the Discovery missions::

For information about NASA's overall mission operations and communications services, contact:

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It is NASA policy that space missions receiving funding from NASA comply with all international and United States regulations, standards, and agreements. Such regulations and standards include those promulgated by:

Information about the ITU and NTIA, regulations can be obtained from NASA's Spectrum Management Office at the Glenn Research Center or by consulting References 2 and 3. Recommended CCSDS standards applicable to DSN, Ground Network, or TDRSS support can be obtained from Reference 4, the CCSDS home page. Recommendations of the SFCG are available in Reference 5.

1.6.1 TMOD Services Interface

1.6.2 X-Band and k_A -Band Communications

1.6.3 Use of File Transfer Protocols

To improve DSN station utilization efficiency, all DSN users should use the CCSDS File Transfer Protocol (FTP), known as CFDP, to download telemetry data from their spacecraft to Earth.

Those proposals choosing to employ the DSN and which are selected for the concept study should work with JPL/TMOD in order to assess the schedule load which the proposed investigation will impose on the DSN. The results of this assessment should be included in the concept study report.

Most proposed Discovery missions will operate in deep space which forces missions to operate with restricted communications bandwidths as compared to missions orbiting close to Earth. In developing their mission concept, proposers usually perform trade-offs among the elements of the end-to-end data system. The elements include instrument format design, flight data system, the space communications features, and the several elements of the ground data system. The integrated contact time and the contact frequency of the spacecraft with the DSN are typically important parameters in these trade studies.

2.1 Deep Space Network

Testing to establish compatibility between the spacecraft's Radio Frequency Subsystem (RFS) and DSN stations is available at the Development Test Facility (DTF-21) at JPL in Pasadena or by using the Compatibility Test Trailer (CTT) at a remote site. RFS compatibility testing is highly recommended and should be completed about one year prior to launch.

The DSN 11-meter stations are designed for space-based very long baseline interferometry (SVLBI) and operate at 7, 8, and 15 GHz. SVLBI missions are characterized by high data rates and nearly continuous Earth station support requirements. Because of their limited capability, 11-meter stations are priced at a flat rate.

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2.3 Multiple Spacecraft Supported by a Single Antenna - Fee Reduction

There are a few constraints. First, only a single uplink frequency can be transmitted. In most cases, this means that only one spacecraft in the cluster can operate in the two-way coherent mode. The remainder must be in a one-way mode. Second, multiple, independent receivers are required at the Earth station. This sets a practical limit on the number of spacecraft that can be served simultaneously to 4. Third, ranging and two-way coherent Doppler data can only be obtained from the single spacecraft in a two-way mode.

If this situation applies and the constraints are acceptable, then it may be possible to reduce the Antenna cost for all spacecraft operating in this mode. To calculate the cost, first compute the *Aperture Fee* using equation 2-1 above. Thereafter, apply the correction factor according to the formula:

$$AF' = (0.75) AF \quad (2-2)$$

where:

AF' = weighted *Aperture Fee* per hour of use when 2 or more spacecraft simultaneously share the same antenna.

The reduced price, AF' , reflects the reduced capability available as a result of sharing. It assumes that the uplink and ranging capabilities will rotate through all spacecraft on a substantially equal basis.

2.4 Compatibility Testing Cost

TMOD encourages prelaunch compatibility testing as a means to eliminate post launch anomalies and expensive troubleshooting. TMOD maintains a facility known as the Development Test Facility (DTF-21) in Pasadena, California. Except for the high power transmitter and low noise-receiving amplifier, which are not included, DTF-21 is configured much like an operational DSN Earth station.

Approximately one year prior to launch, projects should to bring their Radio Frequency Subsystems (RFS) to DTF-21 for testing. Testing requires approximately two weeks and includes such items as RF compatibility, data flow tests, and transponder calibration.

2.5 Cost Calculations

As a minimum, proposals should contain the set of telecommunications parameters in Table 2-1. While proposers may or may not wish to use a tabular format, the required parameter values should be supplied in a clear, concise, and readily apparent form. Table 2-2 is an example of such a form containing 20 parameter values in only 1/3 of a page.

The 11-meter station is charged at a flat rate of $0.2R_B$ irrespective of the number of hours that they are used each week.

Table 2-1: Telecommunications Parameters and Definitions

Parameter	Units	Description
Maximum S/C Distance	Km	Maximum spacecraft-earth station distance during primary mission.
Encounter 1 Distance	Km	Maximum spacecraft-earth station distance during first encounter.
Encounter 2 Distance	Km	Maximum spacecraft-earth station distance during second encounter.
Encounter N Distance	Km	Maximum spacecraft-earth station distance during Nth encounter.
Uplink Transmitter Power	Watts	Earth Station Transmitter Output.
Uplink Frequency Band	GHz	Proposed earth-to-space frequency band expressed in GHz.
Uplink Transmitting Antenna	DBi	Gain (or name) of earth stations transmitting antenna (e.g., 34M BWG).
S/C Receiving Antenna Gains	DBi	Gains of all spacecraft receiving antennas.
Telecommand Data Rate	b/s	Maximum desired telecommand data rate.
Telecommand Bit-Error-Rate	-	Required telecommand Bit-Error-Rate (BER).
S/C Receiver Bandwidth	Hz	S/C Receiver's phase-locked-loop threshold bandwidth (2 Blo).
Turnaround Ranging	Yes/No	Statement whether turnaround ranging is required.
SC Transmitting Power	Watts	S/C Power amplifier Output.
Downlink Modulation Format	Name	Format name (e.g., PCM/PM/Bi-Û, PCM/PSK/PM, BPSK, QPSK, etc.).
Downlink Frequency Band	GHz	Proposed space-to-earth frequency band expressed in GHz.
S/C Transmitting Antenna	DBi	Gains of all spacecraft transmitting antennas.
Downlink Receiving Antenna	DBi	Gain (or name) of earth station receiving antenna (e.g., 34M BWG) .
Telemetry Data Rate	b/s	Maximum desired telemetry data rate.
Telemetry Coding	Name	Telemetry code (e.g., convolutional, Reed-Solomon, concatenated, etc.).
Telemetry Bit-Error-Rate	-	Required telemetry Bit-Error-Rate (BER).

Table 2-2: Sample Table for Inclusion in Proposal

Parameter	Value	Parameter	Value
Maximum S/C Distance (km)		S/C Receiver Bandwidth (Hz)	
Encounter 1 Distance (km)		Turnaround Ranging (Yes/No)	
Encounter 2 Distance (km)		S/C Transmitting Power (Watts)	
Encounter N Distance (km)		Downlink Modulation Format (Name(s))	
Uplink Transmitter Power (Watts)		Downlink Frequency Band (GHz)	
Uplink Frequency Band (GHz)		S/C Transmitting Antenna Gains (dBi)	
Uplink Transmitting Antenna Gains (dBi)		Downlink Receiving Antenna Gain (dBi)	
S/C Receiving Antenna Gains (dBi)		Telemetry Data Rate (b/s)	
Telecommand Data Rate (b/s)		Error Detecting-Correcting Code (Name)	
Telecommand Bit-Error-Rate		Telemetry Bit-Error-Rate	

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DSN SUPPORT SUMMARY	
Total Station Cost:	
TMS Manager Cost:	
NOPE Cost:	
Total DSN Hours:	
Total Support Cost:	

Service Description	Year of Service	TMS Manager		NOPE		Instructions
		Work-Yrs	Cost (\$K)	Work-Yrs	Cost (\$K)	
						1. Complete summary table in upper left hand corner.
						a. Specify <i>Launch Year</i> as 4-digits (e.g., 2002).
						2. In main table above, <u>each line</u> must:
						a. Specify DSN antenna size (e.g., 11, 26, 34, 70).
						1) Specify 34M as either 34BWG or 34HEF subnet.
						b. Name a single year using 4-digits (e.g., 2002).
						1) Only one year permitted one each line.
						c. Contain same level of support requirements.
						3. Calibration times added to each pass automatically.
						a. Pre-Cal. = 30 minutes; Post-cal. = 15 minutes.
						4. All costs computed in real-year dollars.
						a. Costs automatically inflated at official NASA Hq. rate.
						5. Enter TMS Mgr. & NOPE WY requirements in table to left.
						a. TMS Mgr. & NOPE costs computed in real-year dollars.
						6. Summary of total costs appear in table at upper right.
						7. Only Government missions handled at this time.
	Total:					

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AMMOS offers users a selection of services and tools for spacecraft command and control, data reduction and analysis, and navigation. TMOD services are integrated, and certain DSN services may be a prerequisite to obtaining AMMOS value-added services. The main AMMOS elements are located at JPL; however, specific subsystems may also be placed at user sites. Proposals should identify specific AMMOS services listed in Reference 7 required for mission support and the costs for each service over the life of the mission.

Because each mission is unique, it is difficult to provide a priori tool prices. Generally, AMMOS personnel need to confer with cognizant project personnel to determine specific tool requirements. Thereafter, it should be possible to quote a price for the product. If a tool's specification can be completed by the end of Phase B, work can commence at the start of Phase C/D so that the tool will be available at launch.

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Prospective users of SOMO and TMOD facilities can obtain additional information from the following documents:

1. *SOMO Services Catalog*, NASA Space Operations Management Office, Lyndon B. Johnson Space Center, National Aeronautics and Space Administration, Code TA, 2101 NASA Road 1, Houston, Texas 77058. **Copies of the document are available at:** <http://www.jsc.nasa.gov/somo/>.
2. *Radio Regulations*, International Telecommunications Union, Geneva, Switzerland.
3. *Manual of Regulations and Procedures for Federal Radio Frequency Management*, National Telecommunications & Information Administration, U.S. Department of Commerce, Washington D.C., Latest Edition. **Information is available at:** <http://www.ntia.doc.gov/osmhome/redbook/redbook.html>
4. Consultative Committee for Space Data Systems (CCSDS). Blue Books published by the CCSDS Secretariat, NASA Headquarters, Washington D. C. 20546. **Copies of CCSDS Recommendations are available at:** http://www.ccsds.org/blue_books.html.. CCSDS CFDP information is available at: http://www.ccsds.org/red_books.html.
5. *Handbook of the Space Frequency Coordination Group*, ESA Frequency Manager and SFCG Secretariat, European Space Agency Headquarters, 8-10 Rue Mario Nikis, 75738 Paris, France. **Copies of the document are available at:** <http://sfcg.lerc.nasa.gov/>
6. *TMOD's Mission Operations and Communications Services (A Handbook for Preparing Proposals)*, Telecommunications and Mission Operations Directorate, Jet Propulsion Laboratory, Pasadena, California, **Latest Edition**. **Copies of the document are available at:** <http://deepspace.jpl.nasa.gov/advmis>
7. *Telecommunications and Mission Operations Directorate Services Catalog*, Organization 920, Jet Propulsion Laboratory, Pasadena, California, V 6.0 (or later edition), September 1998. <http://jpl-madb.jpl.nasa.gov/srp/catalogHTML/CATALOG-FS.htm>